Topography Correction for Airborne Synthetic Aperture Radar

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Abstract: Airborne synthetic aperture radar (SAR) is an essential tool for modern remote sensing applications. The motion compensation (MOCO) in SAR processing is usually carried out by assuming a reference level to compute the range displacements and phase corrections to apply to each received echo. This means that phase histories of targets at heights different from the reference level cannot be matched accurately, which might yield several effects in the final compressed image. Topography correction for airborne SAR accommodates topography variations during SAR data processing, using an external digital elevation model (DEM). The aperture-dependent MOCO is compensation the phase error of all targets before azimuth compression, resulting in an enhanced image quality. In this paper, analysis the effect of topography variations in focused image (after range and azimuth compression). Then presented an efficient way to use the information given by an external DEM to take into account the motion of the aircraft along the whole synthetic aperture is presented. Finally, real simulated-data experiments show that the proposed approach is appropriate for highly precise imaging of airborne SAR.

Keywords: Airborne synthetic aperture radar, Motion Compensation, aperture-dependent MOCO.

1-Introduction

Synthetic aperture radar (SAR) is a remote sensing system to generate high-resolution microwave images of the observed scene. It is mounted on-board a platform, such as aircraft, satellite or unmanned aerial vehicle (UAV), which is supposed to fly at a constant velocity along a nominal trajectory. For SAR systems, motion is a solution as well as a problem. The problem arises from trajectory deviations and instability of the platform velocities, which not only cause serious image blurring, but also geometric distortion of the SAR imagery. The motion errors can be obtained from the synchronous measurements of navigation systems, and motion compensation (MOCO) is then performed subsequently. The navigation measurements usually provide only coarse MOCO, such as removal of the non-systematic range cell migration (NRCM) and partial phase errors [1-4]. Autofocus approaches are subsequently applied to estimate the residual motion errors. A method presented in [5] is based on the combination of range alignment and phase gradient autofocus (PGA) [6]; the NRCM is corrected by the range alignment techniques with the inertial navigation system (INS) measurements, and the phase error is estimated by PGA. In the raw-data-based approaches, autofocus techniques are employed to estimate motion errors adaptively. PGA is one of the most popular autofocus algorithms. Recent improvements in the PGA algorithms, including the quality PGA (QPGA) and the weighted phase estimation by the weighted PGA (WPGA), are presented in [7, 8], respectively. The NRCM MOCO strategy suffices for a flat terrain and for resolutions in the order of 1 m [9]. While the topographic variation within the observed scene and azimuth dependence of the motion error are considered, a new MOCO strategy, referred to as precise topography- and aperture-dependent MOCO, is proposed in [9-11].
In this paper, we analyzed the effect of topography variations of illuminated area in focused image. That we get after range and azimuth compression. Then explain proposal method aperture-dependent MOCO for topography correction that applied after second order MOCO and before azimuth compression.

We organize this paper as follows. In Section 2, the error due to topography variations derived that affected on focused image. Then, Section 3 used the aperture-dependent MOCO is compensation the phase error of all targets before azimuth compression. In Section 4, the simulated data is used to validate the proposed method. Finally, conclusions are drawn in Section 5.

2. Effect of topography variations on focused image

The Airborne-SAR data acquisition geometry is shown in Fig. 1, where the linear straight line (Y-axis) denotes the nominal track, and the curve represents the real or actual track. In the ideal case, the antenna phase center (APC) of the radar moves along the nominal path at a constant velocity \( V \). However, owing to the displacement of the real track from the nominal one, additional range error from radar to target arises. Assume that the pulse repetition interval is \( T_r \). APCs are located at a constant interval of \( V T_r \) along the Y-axis direction with a reference height of \( H \). The actual and ideal APC positions at the slow time \( t_a \) are 

\[
[X(t_a), Y(t_a), Z(t_a)] \quad \text{and} \quad [0, V t_a, H],
\]

respectively. The motion error vector, defined as the displacement between the actual and nominal paths, is 

\[
d(t_a) = [\Delta x(t_a), Y(t_a) - V t_a, \Delta z(t_a)]
\]

where \( \Delta x(t_a) \) and \( \Delta z(t_a) \) denote the motion displacements in the normal plane, i.e., the cross-track motion errors. The position of the illuminating antenna is now completely described by the azimuth coordinate \( y' = V t_a \) and by the vector \( d(t_a) \), the platform displacements from the nominal track. The correction is made assuming a constant reference height plan at \( z_o \), which is the reference plan. Considering a scatter point at \( P(x, y, z) \) and in the unsquinted observation mode (broadside), the position of that point in reference plan is \( P_o(x_o, y_o, z_o) \). Then, the instantaneous range from the nominal APC position \( y' \) to the scatter \( P_o \) is:

\[
R_n(t_a; r, y) = \sqrt{x_o^2 + (y' - y_o)^2 + [z_o - H]^2} = \sqrt{(y' - y_o)^2 + r^2}
\]

where \( r = \sqrt{x_o^2 + (z_o - H)^2} \) is the distance of the closest approach with respect to the nominal track.

The instantaneous range from the actual APC position \( A \) to the scatter \( P \) is

\[
R(t_a; r, y) = \sqrt{(x - \Delta x)^2 + [y - (y' + \Delta y)]^2 + [(z_o + \Delta h) - H - \Delta z]^2}
\]

\[
= \sqrt{(x_o + x_{sh} - \Delta x)^2 + [(y_o + y_{sh}) - (y' + \Delta y)]^2 + [(z_o + \Delta h) - H - \Delta z]^2}
\]

\[
= R_n(t_a; r, y) + \delta R(t_a; r, y)
\]

where \( \delta R(t_a; r, y) \) is the slant range error, \( \Delta h \) is the difference between actual height and reference high tog scattered, and \( x_{sh} \) is the shift in X-axis due to topography variation, and
\( y_{\Delta h} \) is the shift in Y-axis due to topography variation. The slant range error can be approximately as follows:

\[
\delta R(t_a;r,y) = R(t_a;r,y) - R_0(t_a;r,y) \\
\approx \frac{(V_{t_a} - y_a)}{R_0(t_a;r,y)} \Delta y(t_a) - \frac{x_o}{R_0(t_a;r,y)} \Delta x(t_a) - \frac{(z_o - H)}{R_0(t_a;r,y)} \Delta z(t_a) + \frac{\Delta x(t_a)^2 + \Delta z(t_a)^2 + \Delta y(t_a)^2}{2R_0(t_a;r,y)} + \left( \frac{(y'' - y_a) + \Delta y}{R_0(t_a;r,y)} \right) y_{\Delta h} \]

(3)

Fig. 1 SAR system geometry in the presence of trajectory deviation and topography variations.

Let \( \theta \) denote the off-nadir angle, i.e., the look angle associated with the scatterer \( P \), and \( \alpha \) is the instantaneous squint angle as a function of azimuth position. That can be written as following:

\[
\sin \alpha = \frac{(V_{t_a} - y)}{R_0(t_a;r,x)} , \quad \cos \alpha = \frac{r}{R_0(t_a;r,x)} , \quad \sin \theta = \frac{x}{r} , \quad \cos \theta = \frac{(z_o - H)}{r} \]

\[-\theta_a / 2 < \alpha < \theta_a / 2 \]

(4)

where \( \theta_a \) is the azimuth beamwidth. Substituting Eq. (4) into Eq. (3), and ignoring the higher order terms, the range displacement can be approximated as
\[ \delta R(t_a; r, x) \approx \sin \alpha \Delta y - \cos \alpha (\sin \theta \Delta x + \cos \theta \Delta z) \]
\[ + [\cos \alpha \sin \theta] x_{\text{sh}} - [\sin \alpha] y_{\text{sh}} + [\cos \alpha \cos \theta] \Delta h \]
\[ = \delta R_{\text{ref}}(t_a; r, x) + \delta R_{\text{topo}}(t_a; r, x) \]

where \( \delta R_{\text{ref}}(t_a; r, x) \) and \( \delta R_{\text{topo}}(t_a; r, x) \) are range error related to reference plane and topographic variation are shown in Eq. (6) and Eq. (7), respectively.

\[ \delta R_{\text{ref}}(t_a; r, x) = \sin \alpha \Delta y - \cos \alpha (\sin \theta \Delta x + \cos \theta \Delta z) \] (6)

\[ \delta R_{\text{topo}}(t_a; r, x) = -\sin \alpha y_{\text{sh}} + \cos \alpha (\sin \theta x_{\text{sh}} + \cos \theta \Delta h) \] (7)

In Eq. (6) and Eq. (7) the first terms in are caused by the along-track motion error and the second terms are the cross-track error, and the last four terms is the effect of topographic variation. Assuming that range compression and range cell migration correction (RCMC) have been applied, the signal in time domain for a given target has the following expression in the unsquinted observation mode (broadside):

\[ s(t_a, r) = A_o \exp \left[ -j \frac{4\pi}{\lambda} \sqrt{r^2 + V^2 (t_a - t_{\omega})^2} \right] \times \exp \left[ -j \frac{4\pi}{\lambda} \varepsilon(t_a, r) \right] \]
\[ \times s_{\text{rc}}(t_a - t_{\omega}) \times s_{\text{rc}} \left( t - \frac{2r}{c} \right) \] (8)

where \( A_o \) is a complex constant, \( t_a \) is the azimuth time, \( t \) is the range time, \( r \) is the closest approach distance, \( V \) is the forward velocity of the platform, \( t_{\omega} \) is the zero-Doppler time position, \( c \) is the speed of light, \( \varepsilon(t_a, r) \) is the uncompensated trajectory for that target, \( s_{\text{rc}}(\cdot) \) is the range compressed envelopes, and \( s_{\text{a}}(\cdot) \) is the azimuth compressed envelopes.

In chirp scaling algorithms, a two-step MOCO is commonly applied [12], where first-order MOCO compensation both envelope and phase for a reference range and height, while second-order MOCO corrects for each range after RCMC and range compression. Therefore, second-order MOCO is carried out multiplying Eq. (8) by a complex function containing the residual range-dependent correction \( \varepsilon_0(t_a, r) \), where subscript \( z_0 \) means the correction is made assuming a constant reference height. If the introduced term is equal to the error \( \varepsilon(t_a, r) \), MOCO is applied correctly, i.e., the height of the target is equal to the reference one used during second-order MOCO. However, this is normally not the case if strong topography variations are present in the scene. Therefore, a phase error remains along the phase history of the target, which, after azimuth compression, yields to phase errors and both degradation and displacement of the impulse response along azimuth direction. Consequently, although conventional MOCO has been applied, the error depends on the topography, making azimuth compression still space-variant. The main problem to overcome is the fact that for a given pulse, it is not possible to correct for more than one height. The subaperture approach presented in Section 3 expounds a solution to this problem.

The phase offset value due to second-order MOCO mismatch can be evaluated analytically for the maximum of the impulse response. Assuming time domain azimuth compression, i.e., a cross-correlation, the expression for instant \( t_{\omega} \) is
where $L_{\text{SAR}}$ is the length of the synthetic aperture in seconds. The integral in Eq. (9) should have a zero phase value. With the use of an external DEM, one could think of computing the integral in Eq. (9) for each pixel of the image, and correct the phase. The next section expounds a solution that avoids all harmful effects by modifying the phase history of targets accurately before azimuth compression.

### 3. Topography-Dependent Motion Compensation with Subaperture

Topography-dependent motion compensation started after conventional second-order MOCO using the height information of an external DEM. This solution has the drawback that the correction is only applied at one referable height, which could be the mean height of the antenna footprint for that pulse, and which is not able to accommodate for other heights. The system parameters are listed in Table 1 and a synthetic aperture of ~350 m in midrange. If the observed scene had strong topography variations, the correction would be insufficient.

<table>
<thead>
<tr>
<th>Nominal height ($H$)</th>
<th>12000 m</th>
<th>Chirp duration ($T_p$)</th>
<th>50 µs</th>
</tr>
</thead>
<tbody>
<tr>
<td>Midrange coordinate ($R_o$)</td>
<td>18275 m</td>
<td>Chirp bandwidth ($B_r$)</td>
<td>180 MHz</td>
</tr>
<tr>
<td>Wavelength ($\lambda$)</td>
<td>1.875 cm</td>
<td>Range pixel dimension ($R_c = c/2f_s$)</td>
<td>50 cm</td>
</tr>
<tr>
<td>Platform velocity ($V$)</td>
<td>208 m/s</td>
<td>Azimuth pixel dimension ($A_v = V/PRF$)</td>
<td>20.8 cm</td>
</tr>
<tr>
<td>Sampling Frequency ($f_s$)</td>
<td>300 MHz</td>
<td>Pulse Repetition Frequency ($PRF$)</td>
<td>1000 Hz</td>
</tr>
</tbody>
</table>

The algorithm proposed in this section allows for an angle-accommodation to follow a similar principle than the one presented in [13-15]. The distinction between targets in azimuth by Doppler beam sharpening (DBS) are shown in Fig. 2.
The max number of samples $N_s$ depends on the concept of Doppler Beam Sharpening (DBS). DBS technique takes the azimuth fast Fourier transform process to construct DBS sub-images right after range pulse compression. Fig. 2 shows the geometric relationship between the point target and airborne-SAR in slant range plane. In Fig. 2, airborne-SAR makes a uniform rectilinear path along the Y-axis with the velocity $V$, $P$ is the scatter point target, and $\alpha$ is the instantaneous squint angle as a function of azimuth position. The slant ranges of point target is $r$ and $r + \Delta r$ corresponding to the positions of the airborne-SAR $A_1$ and $A_2$ respectively. The echo of a point target of $P$ in Fig. 2 can be obtained as follows:

$$s_A(t_a, r) = \exp \left[ -j \frac{4\pi}{\lambda} \left( \sqrt{r^2 + V^2 t_a^2 - 2rV t_a \cos \left( \frac{\pi}{2} - \alpha \right)} \right) \right]$$

$$\approx \exp \left[ -j \frac{4\pi}{\lambda} \left( r - V t_a \sin(\alpha) + \frac{V^2 t_a^2 \cos^2(\alpha)}{2r} \right) \right] t_a \in \left[ -\frac{T_c}{2}, \frac{T_c}{2} \right]$$

(10)

where $T_c$ is the integration time in the azimuth direction. To realize the coherent summation, the following inequality is indispensable, which means that the last phase item of $s_A(\cdot)$ is less $\pi / 2$

$$\frac{4\pi V^2 \cos^2(\alpha)}{\lambda} \left( \frac{T_c}{2} \right)^2 \leq \frac{\pi}{2}$$

(11)

Eq. (11) is the choosing principle of the coherent processing time in the azimuth direction; the result is the restriction of the integration time in the azimuth direction as follows:

$$T_c \leq \frac{1}{V \cos(\alpha)} \sqrt{\lambda r} \approx \frac{\sqrt{\lambda r}}{V}$$

(12)

Then constrain of the number of samples can be computed as follows:
Due to the parameters of Table 1, the number of samples must meet $Ns \leq 90$ sample in azimuth, satisfying constrained of the coherent summation in azimuth. The number of samples $Ns = 64$ is used here; the resolution during MOCO along azimuth dimension is about 5.4 m ($L_{SAR}/Ns$), which allows for an accurate accommodation of topography variations.

\[
Ns \leq \frac{PRF \sqrt{\lambda r}}{V} \quad (13)
\]

Fig. 3 Analysis of point target echo slant range plane

Fig. 4 shows the block diagram of the proposed algorithm. The idea is that selecting small number of samples $Ns$ along azimuth dimension in time domain, and applying azimuth FFT along that same dimension, allows for a time-frequency (or time-angle) dependent correction. With this principle, the authors of [13-15] were able to apply accurate MOCO to low-frequency wide-beam data. A further step is to apply a topography-dependent correction using the same principle. The topography- and aperture-dependent MOCO, the algorithms rely on the well-known time–frequency relation of the azimuth SAR signal. The mapping between time and frequency is expressed by:

\[
f_a(t_a, t_{ao}; r) = \frac{2V}{\lambda} \sin(\beta) = \frac{2V^2}{\lambda} (t_a - t_{ao}) \sqrt{r^2 + V^2 (t_a - t_{ao})^2} \quad (14)
\]

where $\beta$ is the azimuth angle corresponding to an azimuth frequency $f_a$. The relation between azimuth frequencies and azimuth angles through the beam is given by the Doppler formula

\[
\beta(i) = \sin^{-1}\left(\frac{\lambda \cdot f_a(i)}{2V}\right) \quad (15)
\]

where $i$ refers to vector location after azimuth FFT, and $f_a(i)$ is the azimuth frequency related to location or scatter $i$. Eq. (15) means that a certain azimuth frequency corresponds to targets seen from the platform at a certain azimuth angle $\beta(i)$. The conversion between the azimuth angles $\beta(i)$ to azimuth positions $y(i)$ is given by:
\[ y(i) = y_{o_j} + r \tan[\beta(i)] \]  

(16)

where \( y_{o_j} \) is the azimuth center position of subaperture \( i \).

Therefore, knowing the azimuth position and depending on a DEM that has been back geocoded to slant-range geometry, it is possible to know all three coordinates in space of the target \( P_i(x, y, z) \) and, consequently, to compute the correct range error \( \delta R(i) \).

The topography-dependent MOCO algorithm applied after conventional second-order MOCO, because at that stage, range compression has already been applied. Therefore, it is possible to perform the following residual range-dependent phase correction as follows:

\[ H(i, r) = \exp \left[ j \frac{4\pi}{\lambda} (\delta R(i) - \delta R_{ref}(i)) \right] = \exp \left[ j \frac{4\pi}{\lambda} \delta R_{topo}(i) \right] \]  

(17)

where \( \delta R(\cdot) \) is the true range error computed by determine scatter position that satisfy azimuth position and close approach range using DEM, \( \delta R_{ref}(\cdot) \) is the total MOCO correction already applied to the center of segment \( j \), i.e., first and second-order MOCO corrections, and \( \delta R_{topo}(\cdot) \) is the range error due to topographic variation. Notice that Eq. (17) is applied in the range-Doppler domain as depicted in Fig. 4.

After operation of all range in a certain segments, then, the segment is azimuth FFT and stored, before continuing to process the next azimuth segment. After finishing, all segments in azimuth recombine it. Finally, matched filtering as standard stripmap SAR processing is performed the azimuth compression.
Fig. 4 Block diagram of the topography-dependent MOCO algorithm.

4. Simulation Results

An airborne-SAR simulation data is utilized in this part to validate the topography-Dependent MOCO. The heights or altitudes of scene are change from 375m to 677m. The airborne-SAR parameters are shown in Table 1. The scene area contains five point targets, which help to measure the quality of topography-Dependent MOCO.

Fig. 5 shows the position and velocity in N-frame. The nominal and actual positions in east, north and up are shown in Fig. 5 (a), (c), and (e), respectively. Then the nominal and actual velocities in east, north and up are shown in Fig. 5 (b), (d), and (f), respectively. The invariant range error is shown in Fig. 6(a) and the along-track deviation from nominal path is
shown in Fig. 6(b). In Fig. 7(a) and (b) the topographic variation in scene area in two and three dimension are shown, respectively.

Fig. 8(a) is the focused image without MOCO. Then, the focused images with MOCO based on the navigation data (Case 1) are shown in Fig. 8(b). Finally, the focused images with the topography-dependent motion compensation-with subaperture (Case 2) are shown in Fig. 8(c). The rectangle area for PT1-5 in Fig. 8(b) and (c) are comparison in Fig. 9.

Fig. 5 Nominal and actual for position and velocity. (a) East position. (a) East velocity. (b) North position. (b) North velocity. (b) Up position. (b) Up velocity.
Fig. 6 Cross-track and along-track motion errors in real data. (a) Range error. (b) Along-track phase perturbation.

Fig. 7 The altitude of scene area. (a) Two dimension. (b) Three dimension.
Fig. 8 The focused image in different cases. (a) The focused image without MOCO. (b) The focused image with MOCO (Case 1). (c) The focused image with the topography-dependent motion compensation-with subaperture (Case 2).
Fig. 9 Focused in rectangle area in Fig. 8. (a) PT1 in case 1. (b) PT1 in case 2. (c) PT2 in case 1. (d) PT2 in case 2. (e) PT3 in case 1. (f) PT3 in case 2. (g) PT4 in case 1. (h) PT4 in case 2. (i) PT5 in case 1. (j) PT5 in case 2.
Fig. 10 Amplitude response of PT1-5. (a) PT1. (b) PT2. (c) PT3. (d) PT4. (e) PT5.

Table 2 The high power reflector point analysis

<table>
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<tr>
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<th>IRW[m]</th>
<th>PSLR[dB]</th>
<th>ISLR[dB]</th>
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<tr>
<td>PT1</td>
<td></td>
<td></td>
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<tr>
<td>Case 1</td>
<td>3.629</td>
<td>-0.9480</td>
<td>2.2419</td>
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<td>Case 2</td>
<td>0.5428</td>
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<tr>
<td>PT2</td>
<td></td>
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<td></td>
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<td>Case 2</td>
<td>0.5366</td>
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<td>PT3</td>
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<td></td>
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<td>Case 1</td>
<td>4.2945</td>
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<td>Case 2</td>
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<td>PT4</td>
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<td>PT5</td>
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<tr>
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<tr>
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<td>0.5407</td>
<td>-11.5886</td>
<td>-8.0250</td>
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### Table 3 Performance comparison

<table>
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<tr>
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<th>Sharpness</th>
<th>Entropy</th>
<th>Contrast</th>
<th>Dynamic-range (dB)</th>
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<tr>
<td>Fig. 8</td>
<td>without MOCO</td>
<td>3.2136</td>
<td>16.1712</td>
<td>1.6742</td>
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<tr>
<td></td>
<td>Case1</td>
<td>6.2916</td>
<td>16.1161</td>
<td>1.8008</td>
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<td></td>
<td>Case2</td>
<td>42.092</td>
<td>15.8959</td>
<td>15.8959</td>
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<td>Case1</td>
<td>1.4110</td>
<td>11.1770</td>
<td>1.2301</td>
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<td></td>
<td>Case2</td>
<td>25.314</td>
<td>10.8533</td>
<td>3.1698</td>
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<tr>
<td>PT2</td>
<td>Case1</td>
<td>5.4331</td>
<td>11.0641</td>
<td>1.5184</td>
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<td>26.816</td>
<td>10.6618</td>
<td>5.9457</td>
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<tr>
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<td>10.9799</td>
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<td>28.475</td>
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<td></td>
<td>Case2</td>
<td>24.988</td>
<td>10.7616</td>
<td>4.2708</td>
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The focused image and the amplitude responses of PT1-5 are compared in Fig. 9 and Fig. 10 for different cases, respectively. Case1 is the red color plot for MOCO depends on navigation data. Case2 is the blue color plot for the topography-dependent motion compensation-with subaperture. In Fig. 9 and Fig. 10, it is proved that the topography-dependent motion compensation-with subaperture method (case 2) can well compensate the motion error for airborne-SAR due to topographic variation. To make it clearer, The point target analysis for impulse response width (IRW), peak sidelobe ratio (PSLR) and integrated sidelobe ratio (ISLR) measured in azimuth direction of PT1-5 is listed in Table 2.

In Table 2, Fig. 9 and Fig. 10, it can be observed that IRW, PSLR and ISLR of PT1-5 in azimuth direction are improved, and the focusing quality in azimuth for case 2 is largely enhanced in visualization and measurement parameters. In Table 3, it can be observed that the focused image with the topography-dependent motion compensation-with subaperture method (case 2) has the larger image sharpness, higher contrast, bigger dynamic-range, and minimum entropy than the focused image MOCO using navigation data. Therefore, the topography-dependent motion compensation-with subaperture (case 2) is well suitable for the motion compensation of airborne-SARs for topographic variation scene area.

### 5. Conclusions

The focused image affected by topography variation of scene area. For this reason we should take topography variation of illuminated area in consideration and computed accurate phase error for every point in scene area depend on digital elevation model. The strategy used for compensation topography variation is the aperture-dependent MOCO is discussed. The simulation data shows the validity of the topography-dependent motion compensation with subaperture algorithm and proves to be feasible for airborne-SARs to compensation topography variation.

### 6. References


